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FALCONFIX:
A MULTI-MODAL APPROACH
TO FIX COMPUTATION

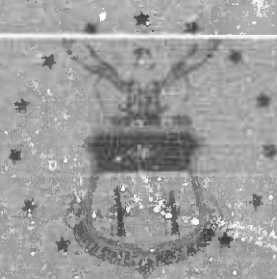
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FINAL REPORT

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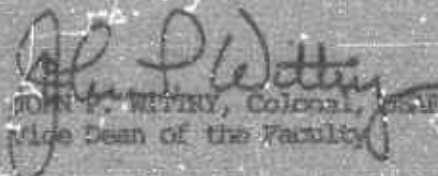
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20. Abstract (continued)

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ABSTRACT

FALCONFIX is a new algorithm for computing the location of an emitter given the bearings from a number of direction-finding stations located worldwide. This approach assumes that each reported bearing may be considered as a sample from a composite normal/uniform distribution, thereby explicitly acknowledging that the bearing may be either "true" or "wild." These distributions are used to create a likelihood function defined over the unit sphere. Non-linear programming techniques are then employed to identify all relative maxima (modes) of this function. The modes (which coincide with concentrations of bearings) are then ordered by value and reported as estimates of the emitter location.

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Chapter 1

INTRODUCTION

Background

A direction-finding system consists of both a net control station and a set of direction-finding sites at widely dispersed geographic locations. When instructed by the net control station, each site tunes its receiver to a specified frequency and, using extremely directional antennas, obtains a bearing on the signal it receives. These bearings are then reported to the net control station where they are processed to estimate the true location of the emitter (obtain a "fix").

Should all the bearings intersect at a single point, the estimate is, of course, obvious. Such a situation rarely occurs in practice since error is induced in the direction of the received signal by both the equipment at each site and by the propagation medium between the sites and the emitter. On the basis of empirical evidence, these errors appear to be approximately normally distributed. Thus, the reported bearings may be thought of as statistical samples taken from normal distributions with unknown true means. Various statistical techniques may be used to estimate these means, thereby providing an estimate of the true location of the emitter.

A particularly vexing aspect of the problem, however, is that of so-called "wild bearings." When a site tunes up on the specified frequency, it might not hear the same signal that other sites are receiving. This can happen both as a result of propagation conditions and of slight differences in the times the various sites listen. In the

latter case a site might report a bearing on the wrong emitter. From a statistical point of view, a wild bearing cannot properly be considered to be a sample from a normal distribution with mean bearing on the true emitter location. Unfortunately, it is impossible to distinguish between bearings taken on the correct emitter from bearings taken on another emitter prior to processing. All fix computation algorithms must deal with this problem in one way or another.

A common method is to consider the problem in two phases. In the first phase, the "most likely" set of true bearings is identified and in the second phase only these bearings are used in making the estimate. The difficulty with this approach is that in deciding which bearings are wild, it is necessary to establish a preliminary estimate of the true location. Then bearings which are not close to this position are considered wild and are rejected. If the preliminary estimate is not close to the true location, it is quite likely that good bearings will be thought to be wild and wild bearings good. It is also possible that enough bearings will be rejected so as to create a "no-fix" situation, i.e., not enough bearings remaining to reliably compute a fix.

Overview of the FALCONFIX Method

The method developed in this paper takes a considerably different approach to the problem--no bearing is ever rejected. Instead, the bearing reported from each site is considered to be a sample from a normal distribution with probability p (a good bearing) and from a uniform distribution with probability $1-p$ (a wild bearing), where p is determined empirically by observing the fraction of good bearings over a period of time. Then a likelihood function, the joint probability distribution of bearing errors assuming this composite normal/uniform distribution, is defined over a spherical earth model. The estimate of true emitter location is then taken as the location which maximizes the likelihood. Because of the nature of each of the bearing error distributions, this likelihood function may have numerous relative maxima (modes), corresponding to points on the earth where intersections of the reported bearings are concentrated.

Non-linear programming and regression techniques are then used to find the location and value of each mode of the likelihood function. The algorithm generates the three most likely modes and orders them according to the probability that they are associated with the true emitter location. The net control station operator may then determine which position to report, on the basis of both the algorithm's ordering as well as other available information.

Finally, the algorithm generates an elliptical confidence region with a specified probability of containing the true position of the emitter.

Chapter 2

MATHEMATICAL DESCRIPTION

Distribution of Bearing Error

The bearing which is reported from a direction-finding site may be taken on either the true target or, inadvertently, on a false target. If it is taken on a true target, we assume that the angular error (ϵ) between the reported bearing and the true bearing is a random variable with a normal distribution having mean zero and variance σ^2 . On the other hand, if the bearing is taken on a false target, no information is conveyed by the reported bearing concerning the bearing to the true target. Therefore, we assume that the angular error is a random variable with a uniform distribution. Further, we assume that the bearing is taken on the true target with probability p and on the false target with probability $(1-p)$; the value for p may be obtained from empirical evidence. For small values of ϵ , therefore, we assume the distribution to be normal and for larger values we assume it uniform. The transition points are designated $\pm c\sigma$, where c is a constant which depends upon both p and σ . Lastly, it is assumed that the probability density function of the bearing error is a continuous function of ϵ . The resulting distribution, which we shall call "composite normal-uniform" is represented by

$$h(\epsilon) = \begin{cases} \frac{1}{K\sqrt{2\pi}\sigma} \exp(-\epsilon^2/2\sigma^2), & -c\sigma \leq \epsilon \leq c\sigma \\ \frac{1}{K\sqrt{2\pi}\sigma} \exp(-c^2/2), & -\pi \leq \epsilon < -c\sigma \text{ or } c\sigma < \epsilon \leq \pi \end{cases}$$

where ϵ and σ are in radians and the normalizing constant K is given by

$$K = [2\phi(c)-1] + 2[\pi-c\sigma] \left[\frac{\exp(-c^2/2)}{\sqrt{2\pi}\sigma} \right]$$

where ϕ is the standard normal cumulative distribution function.

The graph of a typical composite normal-uniform is depicted in Figure 1 below. The fact that the ordinate is constant for values of ϵ greater than $c\sigma$ or less than $-c\sigma$ is extremely significant to the method, as shall become apparent in later sections.

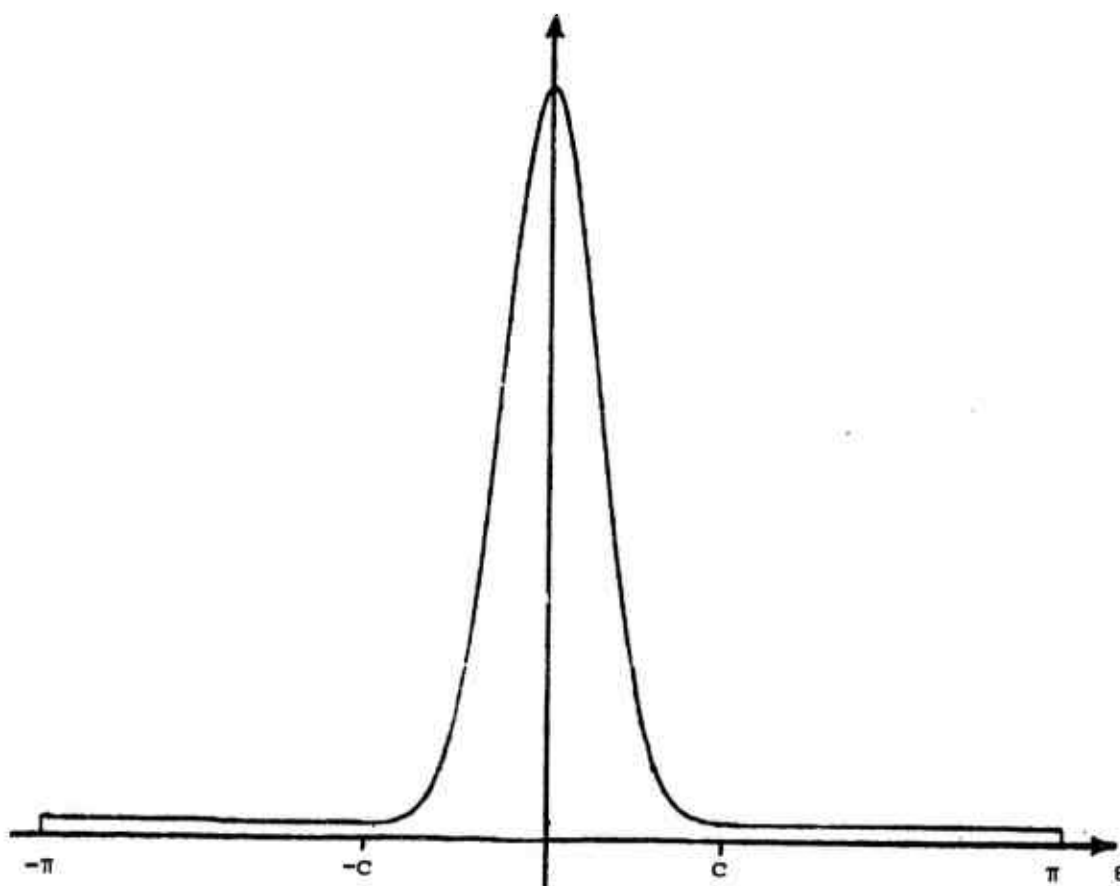


Figure 1. Typical Composite Normal-Uniform PDF

To determine the transition point, we note that

$$p = \frac{1}{K} \int_{-\infty}^{c\sigma} \frac{1}{\sqrt{2\pi\sigma}} \exp(-\epsilon^2/2\sigma^2) d\epsilon$$

$$= \frac{1}{K} [2\Phi(c)-1], \text{ or}$$

$$p = \frac{2\Phi(c)-1}{[2\Phi(c)-1] + 2[\pi-c\sigma] \left[\frac{\exp(-c^2/2)}{\sqrt{2\pi\sigma}} \right]}$$

In Figure 2 below, this relationship between p and σ is plotted for various values of c . Then, given values of p and σ , c can be obtained by interpolating between the curves.

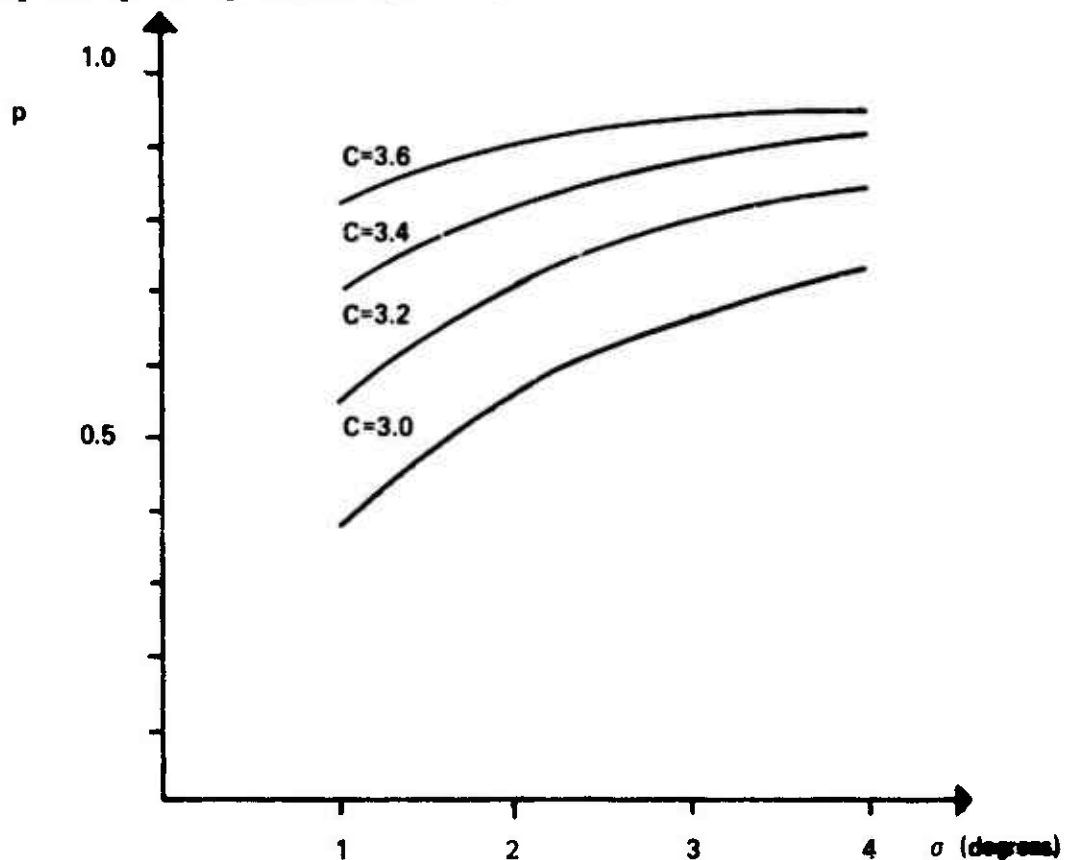


Figure 2. Transition Point Curves

One final approximation for the density function is required to make the procedure more efficient computationally. The ϵ^2 term is difficult to work with using vector methods, so the Maclaurin approximation for $\cos \epsilon$ is used to derive the relationship

$$\epsilon^2 \approx 2(1 - \cos \epsilon).$$

This approximation is correct to within 0.4% for values of ϵ up to 0.2 radians, a value greater than will occur in practice.

Thus, $h(\epsilon)$ is approximated by

$$f(\epsilon) = \begin{cases} \frac{1}{K\sqrt{2\pi}\sigma} \exp \left[\frac{-1}{\sigma^2} (1 - \cos \epsilon) \right], & -\cos \leq \epsilon \leq \cos \\ \frac{1}{K\sqrt{2\pi}\sigma} \exp \left[\frac{-1}{\sigma^2} (1 - \cos(c\sigma)) \right], & -\pi \leq \epsilon < -\cos \text{ or } \cos < \epsilon \leq \pi \end{cases}$$

The Likelihood Function

The likelihood function for the emitter location (ϕ, θ) is defined to be:

$$F(\phi, \theta) = G(\epsilon_1, \epsilon_2, \dots, \epsilon_n) = f_1(\epsilon_1) f_2(\epsilon_2) \cdots f_n(\epsilon_n)$$

where ϵ_i is the angular error between the reported bearing from station i and the true bearing from station i to (ϕ, θ) . Thus ϵ_i is a random variable with density function

$$f_i(\epsilon_i) = \begin{cases} \frac{1}{K_i \sqrt{2\pi} \sigma_i} \exp \left[-\frac{1}{2} \frac{(1 - \cos \epsilon_i)}{\sigma_i^2} \right], & -c_i \sigma_i \leq \epsilon_i \leq c_i \sigma_i \\ \frac{1}{K_i \sqrt{2\pi} \sigma_i} \exp \left[-\frac{1}{2} \frac{(1 - \cos(c_i \sigma_i))}{\sigma_i^2} \right], & -\pi \leq \epsilon_i < -c_i \sigma_i \text{ or } c_i \sigma_i < \epsilon_i \leq \pi \end{cases}$$

Therefore,

$$F(\phi, \theta) = \frac{1}{(2\pi)^{n/2}} \prod_{i=1}^n \frac{1}{K_i \sigma_i} \prod_{i \in W} \exp \left(\frac{-1}{2} \frac{(1 - \cos \epsilon_i)}{\sigma_i^2} \right) \times \prod_{i \notin W} \exp \left(\frac{-1}{2} \frac{(1 - \cos(c_i \sigma_i))}{\sigma_i^2} \right)$$

where $W = \{i: |\epsilon_i| \leq c_i \sigma_i\}$

As will be made clear subsequently, our approach to estimating emitter location will be to find the relative maxima--which we call "modes"--of this function.

Since $\frac{1}{(2\pi)^{n/2}} \prod_{i=1}^n K_i \sigma_i$ is not a function of the ϵ_i , $F(\phi, \theta)$ is

maximized whenever

$$F^*(\phi, \theta) = \prod_{i \in W} \exp \left(- \frac{1}{\sigma_i^2} (1 - \cos \epsilon_i) \right) \prod_{i \notin W} \exp \left(- \frac{1}{\sigma_i^2} (1 - \cos(c_i \sigma_i)) \right)$$

is a maximum. $F^*(\phi, \theta)$, in turn, is maximized whenever its logarithm

$$L(\phi, \theta) = \sum_{i \in W} \frac{-1}{\sigma_i^2} (1 - \cos \epsilon_i) + \sum_{i \notin W} \frac{-1}{\sigma_i^2} (1 - \cos(c_i \sigma_i))$$
 is, and this is the

function to which we shall direct our attention.

Before proceeding with the development of the method, we shall pause to consider exactly what this function $L(\phi, \theta)$ looks like. In Figures 3 through 7 are depicted five examples of a set of bearings and the resulting function L .

In the figure below the case in which two lines of bearing intersect at a single point is illustrated. Notice that the function has a single mode at the point of intersection. (As we shall see later, the FALCONFIX algorithm does not consider two bearing crosses as potential position estimates.)

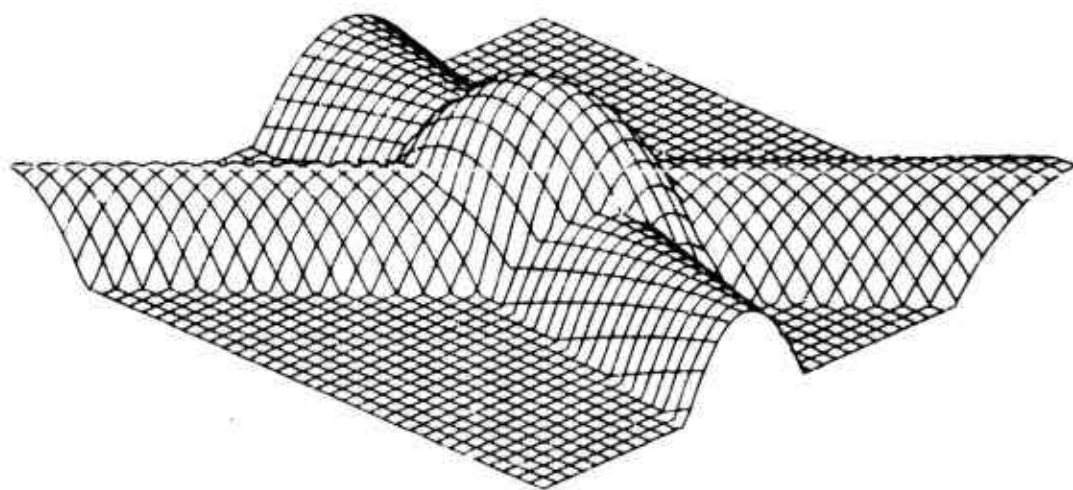


Figure 3. $L(\phi, \theta)$; Two Bearing Cross

In Figure 4 below, the function $L(\phi, \theta)$ in which three bearings intersect at a point is illustrated. Here the mode is somewhat more sharply peaked than in the previous case.

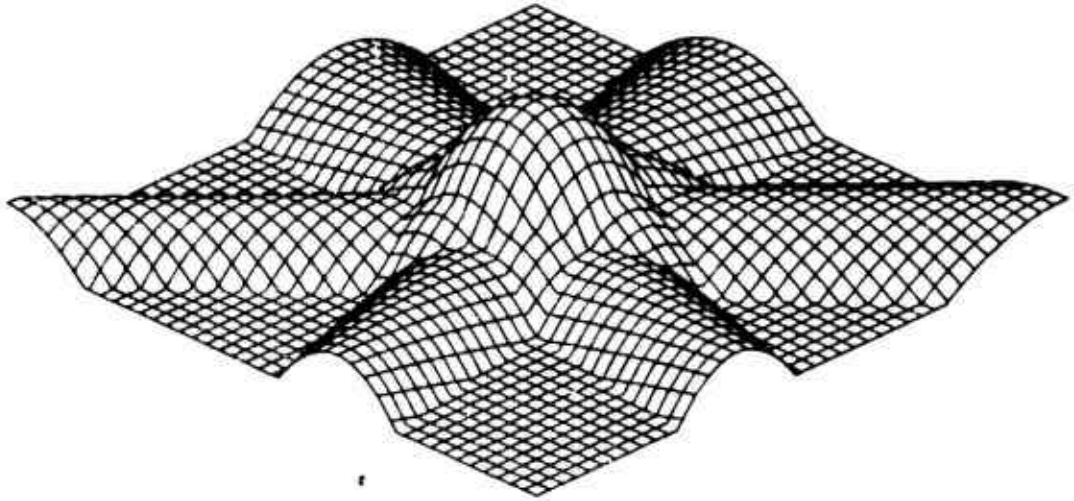


Figure 4. $L(\phi, \theta)$; Three Bearing Case #1

In Figure 5, the three bearings do not meet at a single point, and, in fact, the "cocked hat" is so large compared to the standard deviation of the bearing errors, that the function does not have a single mode in the center. Instead, there are three individual two bearing crosses at each of the intersections.

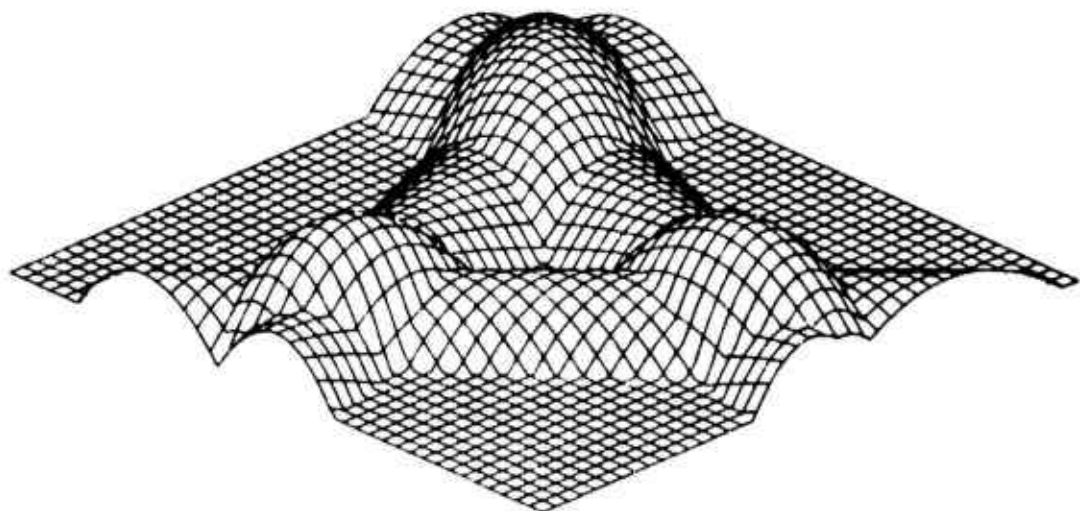


Figure 5. $L(\phi, \theta)$; Three Bearing Case #2

In Figure 6 below, a four bearing case is depicted in which the bearings do not intersect at a single point. Notice that this figure is considerably less regular than the previous figures, and is more typical of what one finds in real cases.

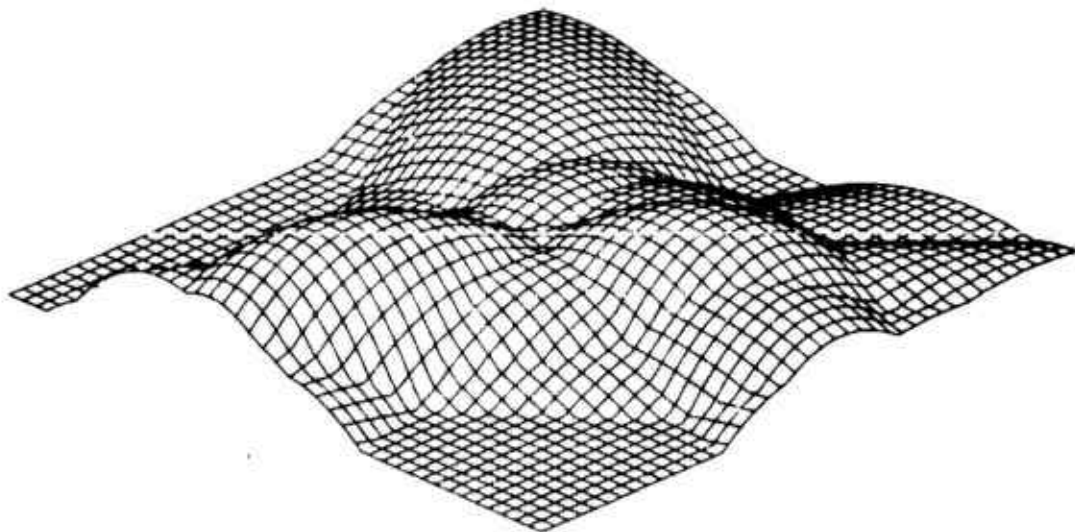


Figure 6. $L(\phi, \theta)$; Four Bearing Case

In Figure 7 the bimodal case is illustrated. In this five bearing example it is not likely that all bearings were taken on the same emitter. Rather, it is much more plausible that either Bearings 1, 2 and 3 are good bearings (with 4 and 5 wild) or Bearings 1, 4 and 5 are good bearings (with 2 and 3 wild). The two modes in the figure correspond to these two possibilities.

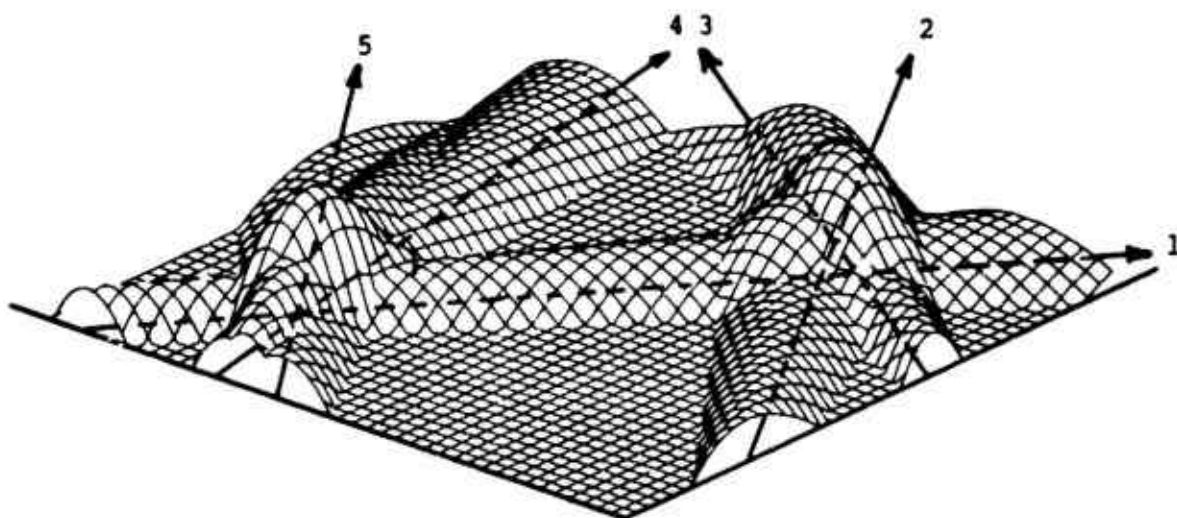


Figure 7. $L(\phi, \theta)$; Bimodal Case

Before going on to the methodology of identifying these modes, it will first be necessary to express each of the ϵ_i in the equation for L explicitly as functions of ϕ and θ , the latitude and longitude of the position estimate. To accomplish this we will rely heavily on the use of vectors as described in the next section.

Vector Representation

A position vector P from the center of the earth to a point (ϕ, θ) on the surface may be represented as

$$P = (\cos \theta \cos \phi)\bar{i} + (\sin \theta \cos \phi)\bar{j} + (\sin \phi)\bar{k}$$

This vector P is depicted in Figure 8 below in a right-hand coordinate system in which \bar{i} points to the prime meridian on the equator and \bar{k} points to the north pole. Thus, east longitude is represented by positive values of θ , west longitude by negative values, north latitude by positive values of ϕ and south latitude by negative values.

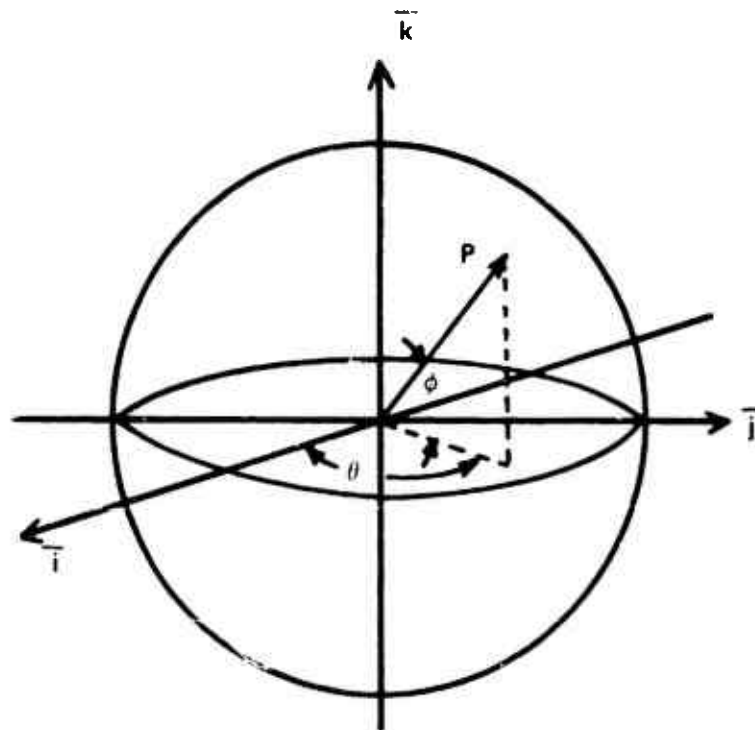


Figure 8. Position Vector

If (ϕ, θ) is the location of the target and (ϕ_i, θ_i) is the location of the i^{th} station, then the corresponding position vectors are:

$$\mathbf{T} = (\cos \theta \cos \phi) \bar{\mathbf{i}} + (\sin \theta \cos \phi) \bar{\mathbf{j}} + (\sin \phi) \bar{\mathbf{k}}$$

$$\mathbf{S}_i = (\cos \theta_i \cos \phi_i) \bar{\mathbf{i}} + (\sin \theta_i \cos \phi_i) \bar{\mathbf{j}} + (\sin \phi_i) \bar{\mathbf{k}}$$

Let β_i be the bearing reported by the i^{th} station. Then in terms of the local coordinate system $(\bar{\mathbf{i}}^*, \bar{\mathbf{j}}^*, \bar{\mathbf{k}}^*)$ in which $\bar{\mathbf{i}}^*$ points east, $\bar{\mathbf{j}}^*$ points north and $\bar{\mathbf{k}}^*$ points to the local zenith, a vector pointing in the direction of β_i may be represented as

$$\mathbf{B}_i^* = (\sin \beta_i) \bar{\mathbf{i}}^* + (\cos \beta_i) \bar{\mathbf{j}}^*$$

which is shown in the figure below.

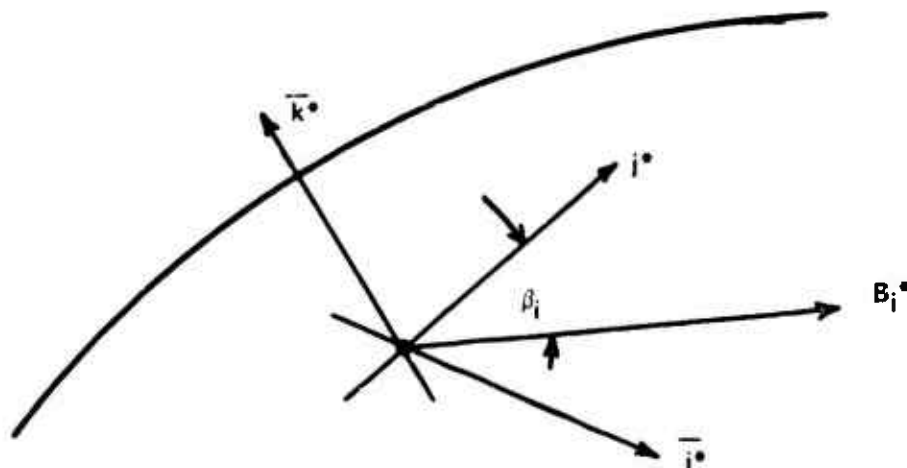


Figure 9. Bearing Vector

To be useful, however, it is necessary to express each bearing vector in the same coordinate system as the \mathbf{T} and \mathbf{S}_i vectors. This is accomplished by rotating the local frame with transformation \mathbf{R}_i so that it coincides with the geocentric frame shown in Figure 8.

Specifically,

$$B_i = R_i B_i^* = A\bar{i} + B\bar{j} + C\bar{k} \text{ where}$$

$$A = -(\sin \phi_i \cos \theta_i \cos \beta_i + \sin \theta_i \sin \beta_i)$$

$$B = \cos \theta_i \sin \beta_i - \sin \phi_i \sin \theta_i \cos \beta_i$$

$$C = \cos \phi_i \cos \beta_i$$

Now, T , the S_i and the B_i are all unit vectors expressed in the same geometric frame.

Then, the vector $\left(\frac{S_i \times T}{|S_i \times T|} \right)$ is a unit vector normal to the "station-target plane," the plane containing S_i and T . Similarly, $\left(\frac{S_i \times B_i}{|S_i \times B_i|} \right)$ is a unit vector normal to the "station-bearing plane," the plane containing S_i and B_i .

Clearly, the angle between these two normal vectors is equal to the dihedral angle between the two planes, which in turn is equal to the angle ϵ_i between the reported bearing and the true bearing from the i^{th} station to the target. This relationship is shown in Figure 10.

From the familiar dot product relationship we have

$$\cos \epsilon_i = \left(\frac{S_i \times T}{|S_i \times T|} \right) \cdot \left(\frac{S_i \times B_i}{|S_i \times B_i|} \right)$$

Since S_i and B_i are orthonormal, it is true that $|S_i \times B_i| = 1$, which means that

$$\cos \epsilon_i = \frac{(S_i \times T) \cdot (S_i \times B)}{|S_i \times T|}$$

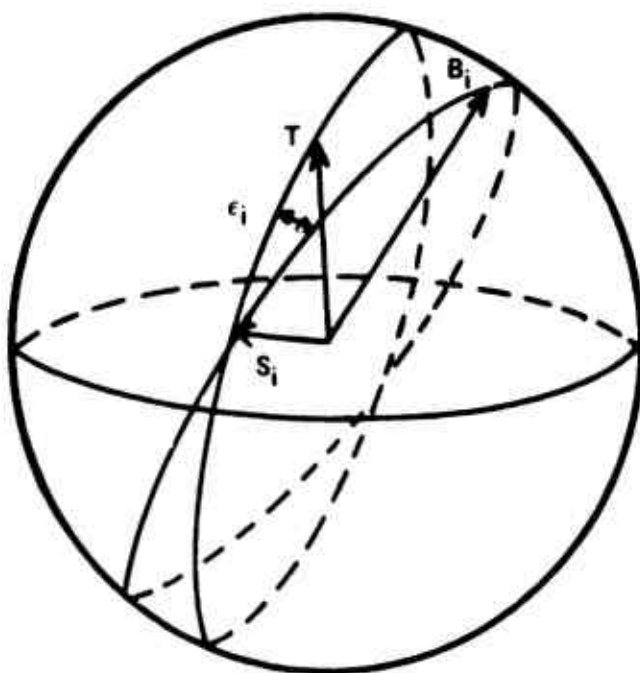


Figure 10. Dihedral Angle

Employing a little-used identity from vector calculus on the numerator of this expression yields

$$(S_i \times T) \cdot (S_i \times B_i) = (B_i \cdot T)(S_i \cdot S_i) - (S_i \cdot T)(S_i \cdot B_i)$$

But $S_i \cdot S_i = 1$ since S_i is a unit vector and $S_i \cdot B_i = 0$ since S_i and B_i are orthogonal.

Thus $(S_i \times T) \cdot (S_i \times B_i) = B_i \cdot T$, which means that

$$\cos \epsilon_i = \frac{B_i \cdot T}{|S_i \times T|}$$

Now $|S_i \times T|$ is equal to the sine of the angle between S_i and T . Since $S_i \cdot T$ equals the cosine of the same angle, we can represent $|S_i \times T|$ as

$$|S_i \cdot T| = \sqrt{1 - (S_i \cdot T)^2}$$

which is somewhat easier to deal with computationally. The relationship

$$\cos \epsilon_i = \frac{B_i \cdot T}{\sqrt{1 - (S_i \cdot T)^2}}$$

will be used in subsequent calculations. Making this substitution, the logarithm of the likelihood function becomes:

$$L(\phi, \theta) = \sum_{i \in W} \frac{-1}{\sigma_i^2} \left(1 - \frac{B_i \cdot T}{\sqrt{1 - (S_i \cdot T)^2}} \right) + \sum_{i \notin W} \frac{-1}{\sigma_i^2} (1 - \cos(c_i \sigma_i))$$

In this equation the c_i , σ_i , B_i and S_i are all constants and T is a function of ϕ and θ . We now have $L(\phi, \theta)$ in a form where it is possible to find the modes by moving T in such a way as to progressively increase the value of L . In the next section we will illustrate exactly how this is done.

Modes of $L(\phi, \theta)$

$L(\phi, \theta)$ can be written in the form:

$$L(\phi, \theta) = \sum_{i \in W} \frac{B_i \cdot T}{\sigma_i^2 \sqrt{1 - (S_i \cdot T)^2}} + K$$

where $K = \text{a constant}$

$$B_i = b_{i1} \bar{i} + b_{i2} \bar{j} + b_{i3} \bar{k}$$

$$S_i = s_{i1} \bar{i} + s_{i2} \bar{j} + s_{i3} \bar{k}$$

$$T = t_1 \bar{i} + t_2 \bar{j} + t_3 \bar{k}$$

$$= \cos \theta \cos \phi \bar{i} + \sin \theta \cos \phi \bar{j} + \sin \phi \bar{k}$$

$$\text{and } |B_i| = |S_i| = |T| = 1$$

A necessary condition for the point (ϕ^*, θ^*) to be a mode (relative maximum) of $L(\phi, \theta)$ is that both

$$\left. \frac{\partial L}{\partial \theta} \right|_{\theta=\theta^*} = 0 \quad \text{and} \quad \left. \frac{\partial L}{\partial \phi} \right|_{\phi=\phi^*} = 0$$

After considerable calculation it can be shown that:

$$\frac{\partial L}{\partial \theta} = \sum_{i \in W} F_i(T) \left(\frac{-b_{i1} \sin \theta \cos \phi + b_{i2} \cos \theta \cos \phi}{B_i \cdot T} + \frac{(S_i \cdot T)(-s_{i1} \sin \theta \cos \phi + s_{i2} \cos \theta \cos \phi)}{1 - (S_i \cdot T)^2} \right), \text{ and,}$$

$$\frac{\partial L}{\partial \phi} = \sum_{i \in W} F_i(T) \left(\frac{-b_{i1} \cos \theta \sin \phi - b_{i2} \sin \theta \sin \phi + b_{i3} \cos \phi}{B_i \cdot T} + \frac{(S_i \cdot T)(-s_{i1} \cos \theta \sin \phi - s_{i2} \sin \theta \sin \phi + s_{i3} \cos \phi)}{1 - (S_i \cdot T)^2} \right)$$

$$\text{where } F_i(T) = \frac{B_i \cdot T}{\sigma_i^2 \sqrt{1 - (S_i \cdot T)^2}}$$

Setting these partial derivatives equal to zero and then solving for θ and ϕ respectively, we obtain:

$$\theta = J(\phi, \theta) = \tan^{-1} \left\{ \frac{\sum_{i \in W} F_i(T) \left(\frac{b_{i2}}{B_i \cdot T} + \frac{(S_i \cdot T) s_{i2}}{1 - (S_i \cdot T)^2} \right)}{\sum_{i \in W} F_i(T) \left(\frac{b_{i1}}{B_i \cdot T} + \frac{(S_i \cdot T) s_{i1}}{1 - (S_i \cdot T)^2} \right)} \right\}$$

and

$$\phi = K(\phi, \theta) = \tan^{-1} \left\{ \frac{\sum_{i \in W} F_i(T) \left(\frac{b_{i3}}{B_i \cdot T} + \frac{(S_i \cdot T) s_{i3}}{1 - (S_i \cdot T)^2} \right)}{\sum_{i \in W} F_i(T) \left(\frac{b_{i1} \cos \theta + b_{i2} \sin \theta}{B_i \cdot T} + \frac{(S_i \cdot T) (s_{i1} \cos \theta + s_{i2} \sin \theta)}{1 - (S_i \cdot T)^2} \right)} \right\}$$

Thus, at the point (ϕ^*, θ^*) these equations will be satisfied. These equations are used to establish the following recursion relationships:

$$\theta_{k+1} = J(\phi_k, \theta_k) \quad \text{and} \quad \phi_{k+1} = K(\phi_k, \theta_{k+1}), \quad k = 0, 1, 2, \dots$$

The FALCONFIX algorithm for estimating (ϕ^*, θ^*) proceeds as follows:

1. Select a starting point (ϕ_0, θ_0) . (The selection procedure is discussed later.)
2. $T_0 = T(\phi_0, \theta_0)$: Initialize the target vector
3. $k = 0$: Initialize k
4. $\delta = \delta_1$: Specify the stopping rule

5. $\theta_{k+1} = J(\phi_k, \theta_k)$: Calculate the next longitude
6. $\phi_{k+1} = K(\phi_k, \theta_{k+1})$: Calculate the next latitude
7. $T_{k+1} = T(\phi_{k+1}, \theta_{k+1})$: Calculate the next target vector
8. If $k > M$ then $\delta = \delta_2$: Relax, the stopping rule if k is large
9. If $|T_{k+1} - T_k| < \delta$ or $k > N$ then go to 12 : stopping rule
10. $k = k+1$: Increment k
11. Go to 5 : Perform the next iteration
12. $(\phi_{k+1}, \theta_{k+1})$ is an estimate for (ϕ^*, θ^*)

Note: $N > M$ and $\delta_2 > \delta_1$

The intersections of pairs of bearings are used as the starting points for the algorithm. Those pairs which intersect at an angle of greater than 20° are considered first—those intersecting at smaller angles are deferred for later consideration. The first check in determining whether an intersection constitutes a valid starting point is if that particular pair is not a subset of a bearing set W from a previously generated mode. If it is, the intersection is rejected since the iteration procedure would yield the same mode. The second check is to determine if at least one additional bearing is contained in the set W at the point of intersection. If not, the intersection is also rejected as a starting point since in this case no iteration will take place and two-bearing modes are not allowable, except in the special case where only two bearings are input.

The following example illustrates the algorithm and the selection of starting points. Suppose the reported bearings are as shown in

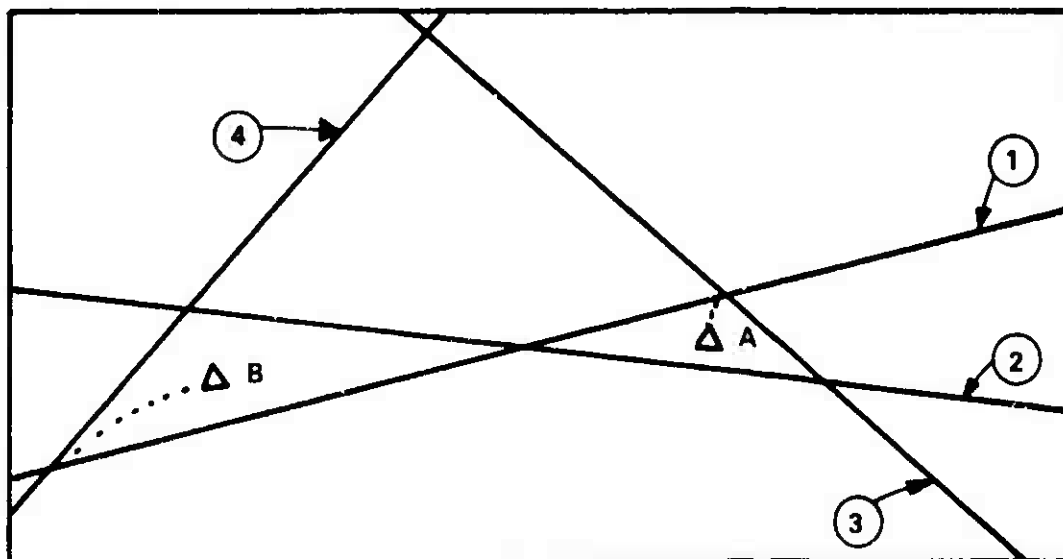


Figure 11. Example of Mode Determination

Figure 11. The intersection (1,2) is temporarily set aside because the angle is too small. (1,3) satisfies this criterion and also that at least one additional bearing, (2), is contained in the bearing set at this point. Therefore (1,3) is a valid starting point. The algorithm commences and after several iterations converges to the point A. Since bearings 1, 2 and 3 are in this bearing set, the intersections (1,2), (1,3) and (2,3) are removed from further considerations. (1,4) is the next candidate and since it meets both criteria it is selected as a starting point. The algorithm begins there and converges to point B. In the same way as before, (1,2), (1,4) and (2,4) are eliminated. (3,4) is the only intersection remaining. It satisfies the angular

criterion, but no other bearing is close enough to be a member of the bearing set at the point of intersection. Therefore (3,4) is eliminated. Since no intersections remain as candidates, all modes have been found and the problem is terminated.

When the algorithm generates more than a single mode, we order them with respect to the probability that they coincide with the true location of the emitter. Since each bearing has a probability p of being a true bearing and for realistic direction-finding systems, this value is greater than 0.5, the mode having the greatest number of forward bearings in its bearing set is considered the most likely. In the case where two or more modes involve the same number of forward bearings, the algorithm uses the degree of dispersion of the bearings as a secondary criterion. Specifically, the average squared perpendicular distance from the position estimate to each line of bearing is computed and the modes are ordered so that the one with the smallest such value is first.

Standard Deviation of Bearing Error

The standard deviation of the bearing error (σ) is used in the calculation of the likelihood function and its partial derivatives. While the particular method used for obtaining σ is not crucial to the FALCONFIX algorithm, for the sake of completeness its computation will be discussed.

The standard deviation is taken to be a function of both the characteristics of the particular receiving site (antenna, equipment, location, etc.) and of the distance between the site and the emitter location. In particular, we assume the multiplicative relationship $\sigma = h(d)\bar{\sigma}$ where $\bar{\sigma}$ is a constant representing the normalized standard deviation of the bearing error from the site and $h(d)$ is an adjustment for distance.

If $s = \sin \delta$, where δ is the central angle between the target vector T and the station vector S, then

$$d = 3444 \sin^{-1}(s) \quad , \quad d \text{ in nautical miles}$$

or

$$s = \sin \left(\frac{d}{3444} \right)$$

The function $h(d)$ below is motivated by the Ross Range Curve, although certain modifications have been made.

$$h(d) = \begin{cases} 60 & , \text{ if } d \leq 60 \\ 60 - 3152.387(s - .0174524) & , \text{ if } 60 < d \leq 120 \\ .174524/s & , \text{ if } 120 < d \leq 600 \\ s + .826352 & , \text{ if } 600 \leq d \leq 5410 \\ 1.826352 & , \text{ if } d > 5410 \end{cases}$$

A graph of $h(d)$ is provided in Figure 12 below.

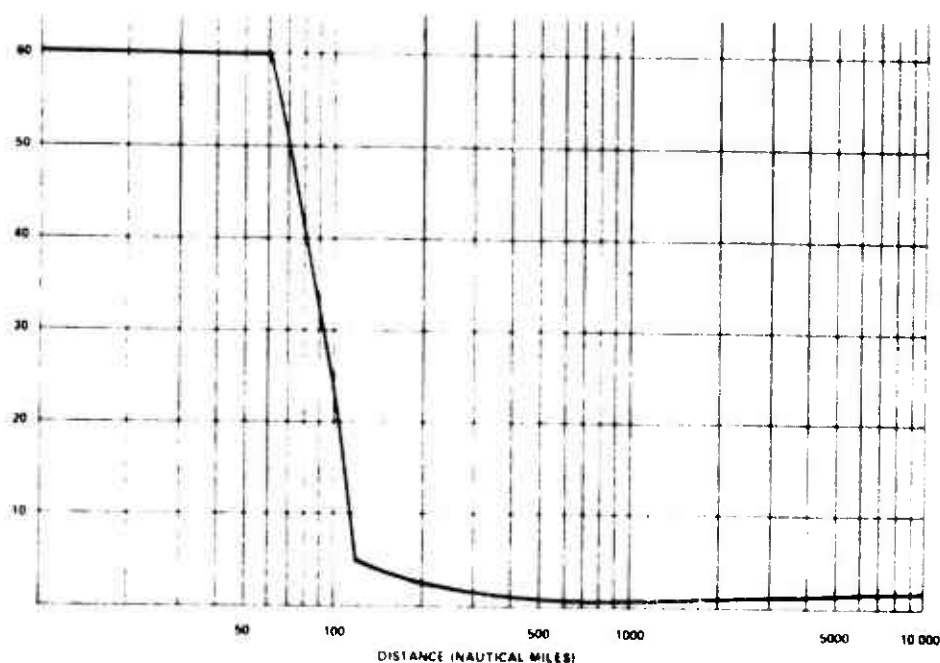


Figure 12. Graph of $h(d)$

The extremely large standard deviation for distances less than 100 nautical miles or so, is essentially an artifice to assure inclusion of bearings from sites which are close to a potential mode; such bearings are almost invariably good bearings. For example, if $\bar{\sigma}$ is equal to 1.0 degrees and if the transition point c (referred to in Figure 2) is set equal to 3.0, then any estimate of emitter location within the "keyhole" in Figure 13 will include the bearing from site s . (In this figure the origin represents the site and the horizontal axis is the direction of the reported bearing). Notice that the bearing is included whenever the site is within 60 nautical miles of the estimate.

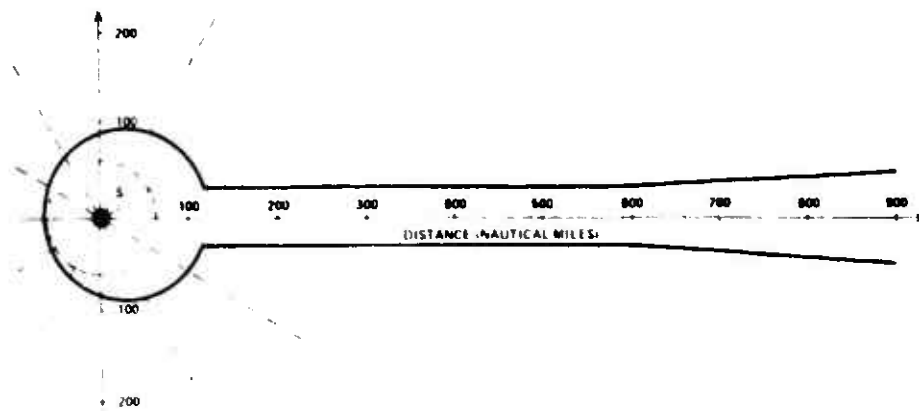


Figure 13. Inclusion Region

The REFIX Procedures

One final procedure is used to refine the position estimates generated by the previously described algorithm. The need for such a termination procedure is two-fold.

- If one of more of the bearings in the included set is from a station which is reasonably close to the position estimate, the artificially large standard deviation associated with such bearings will tend to diminish its rightful impact on the position estimate.
- If the likelihood function happens to be unusually flat in the neighborhood of the mode, it is possible that the stopping rule may terminate the algorithm prematurely, resulting in a reported position estimate which is not coincident with the point at which the relative maximum occurs.

Two different termination procedures have been developed, but both have the following features in common.

- The bearing set is "frozen" to include only those bearings which were included in the final iteration of the algorithm previously described.
- The standard deviations of bearings from stations within 600 nautical miles of the position are modified by changing $h(d)$ to $h^*(d) = 0.75 + d/2400$ where d is the distance in nautical miles.

REFIX/1. In this procedure, after the bearing set has been frozen and the standard deviations of bearings from close-in stations have been modified, the iterative algorithm is employed again.

REFIX/2. This termination procedure uses techniques employed in multiple regression analysis. Most of the computation done here is required for the confidence region calculations which are discussed in the next section. In this procedure a cartesian coordinate system is set up in a plane tangent to the spherical earth model, such that the origin is coincident with the position estimate produced by the iterative procedure. The lines of bearing are assumed to be straight lines in this plane.

A single representative line of bearing is shown in Figure 14 with (x_0, y_0) being the true target location.

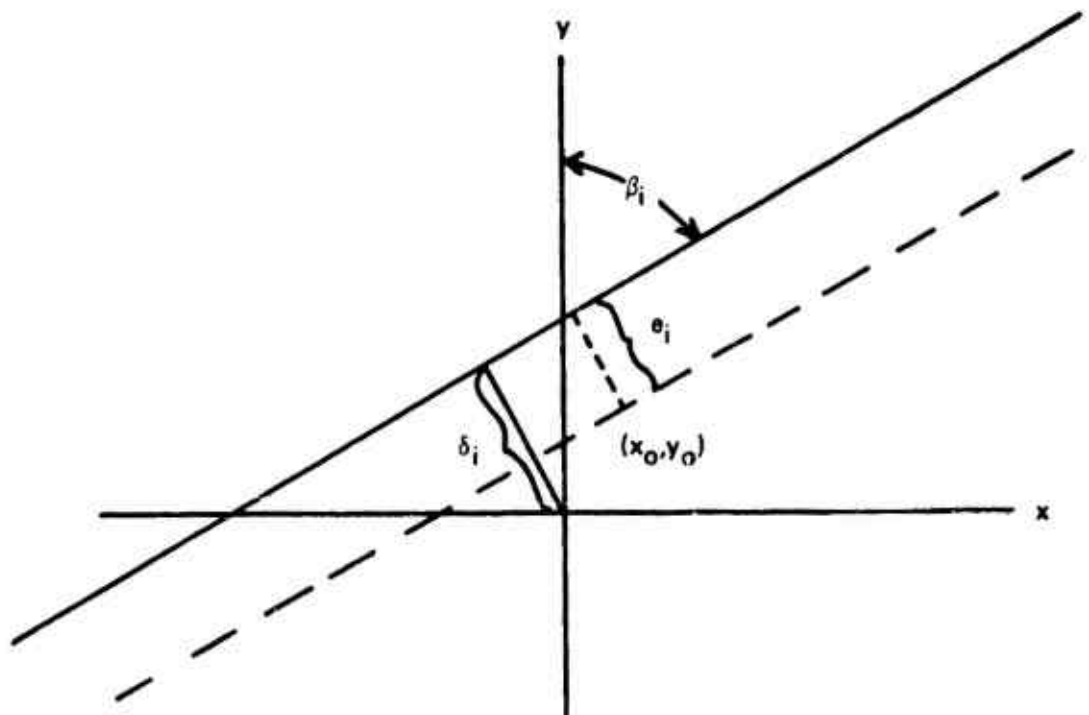


Figure 14. Coordinate System for the REFIX/2 Procedure

From elementary algebra, the equation of the solid line in Figure 14 is $(-\cos \beta_i)x + (\sin \beta_i)y = \delta_i$. The equation of the dashed line through

the true target location (x_0, y_0) is $(-\cos \beta_i)x + (\sin \beta_i)y = \delta_i - e_i$

Then, since (x_0, y_0) is on the dashed line, we have:

$$\delta_i = (-\cos \beta_i)x_0 + (\sin \beta_i)y_0 - e_i$$

This equation is in the form of a general multiple regression model.

Both the β_i and the δ_i can be calculated since the origin is known. The β_i is approximately the bearing from the origin to the station for distant stations and approximately the reported bearing for near stations.

The δ_i is approximately $\gamma_i \sin d_i$ where γ_i is the angular difference between the reported bearing and the bearing from the station to the origin, and where d_i is the distance from the station to the origin.

Also, e_i , the perpendicular distance from the true target location to the bearing, is a normal random variable since $e_i = \epsilon_i \sin d_i$ and ϵ_i (the angular error) is a normal random variable. Furthermore, since the ϵ_i are normal and independent with mean zero and variance σ_i^2 , the e_i will be normal and independent with mean zero and variance $(\sin^2 d_i)\sigma_i^2$.

Thus, the problem reduces to the simple multiple regression problem of estimating \underline{b} when

$$\underline{\delta} = A\underline{b} + \underline{e}$$

where

$$\underline{\delta} = \begin{bmatrix} \delta_1 \\ \vdots \\ \delta_k \end{bmatrix} \quad A = \begin{bmatrix} -\cos \beta_1 & \sin \beta_1 \\ \vdots & \vdots \\ -\cos \beta_k & \sin \beta_k \end{bmatrix}$$

$$\underline{b} = \begin{bmatrix} x_0 \\ y_0 \end{bmatrix} \quad \text{and} \quad \underline{e} = \begin{bmatrix} e_1 \\ \vdots \\ e_k \end{bmatrix}$$

Further, we have that \underline{e} is distributed as a multivariate random variable with $\underline{\mu} = \underline{0}_{k \times 1}$ and diagonal covariance matrix

$$\Sigma = \sigma_o^2 \begin{bmatrix} \sin^2 d_1 w_1^2 & 0 & \cdot & \cdot & \cdot & 0 \\ 0 & \sin^2 d_2 w_2^2 & & & & \\ \cdot & & \cdot & & & \cdot \\ \cdot & & & \cdot & & \cdot \\ \cdot & & & & \cdot & \cdot \\ 0 & \cdot & \cdot & \cdot & \cdot & \sin^2 d_k w_k^2 \end{bmatrix} = \sigma_o^2 W$$

with $\sigma_i^2 = (w_i \sigma_o)^2$ for each station.

Using the above notation, the maximum likelihood estimate of $\underline{b} = (x_o, y_o)$, the true target location, is given by

$$\hat{\underline{b}} = (A^T \Sigma^{-1} A)^{-1} A^T \Sigma^{-1} \underline{\delta}$$

Once $\hat{\underline{b}}$ is available, we update the target vector to give the refined position estimate.

Relative Advantages. Either termination procedure works quite adequately in practice. The REFIX/1 procedure has a slight advantage when the position estimate is very close to one station since the assumption of parallel movement of bearings which is made in REFIX/2 is not really true in this case. On the other hand, REFIX/2 is quicker in some cases since on occasion REFIX/1 requires a number of additional iterations.

The Confidence Region

The calculation of the confidence region follows directly from the REFIX/2 algorithm. The estimate \hat{b} is distributed multivariate normal with mean $\underline{\mu}^* = [x_0, y_0]^T$ and covariance matrix

$$\Sigma^* = (A^T \Sigma^{-1} A)^{-1} = \frac{1}{\sigma_0^2} (A^T W^{-1} A)^{-1}$$

The minimum area $100 (1-\alpha)\%$ confidence region is given by

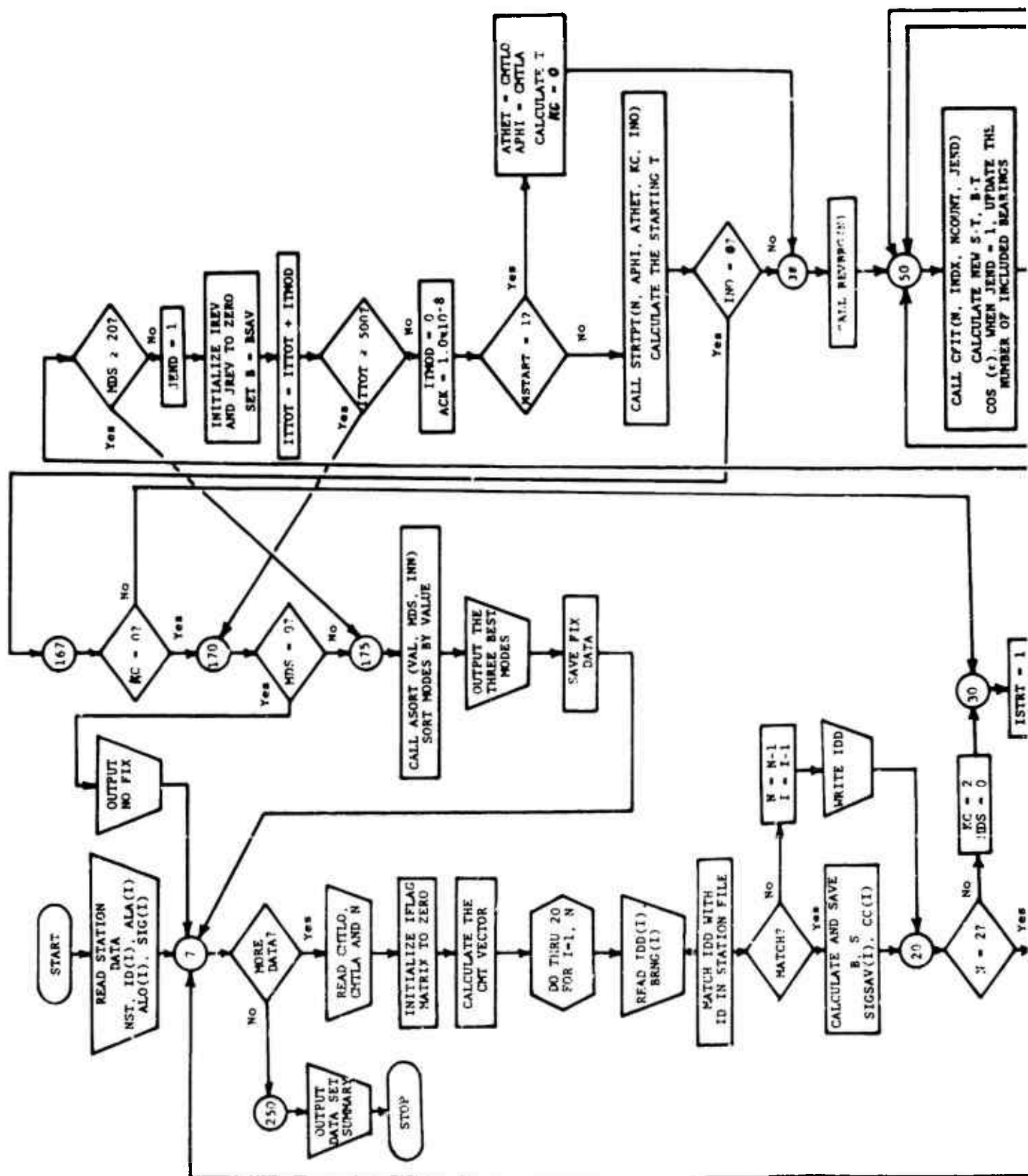
$$\frac{1}{s_0^2} [x, y] \Sigma^* \begin{bmatrix} x \\ y \end{bmatrix} \leq F_{\alpha, 2, n-2}$$

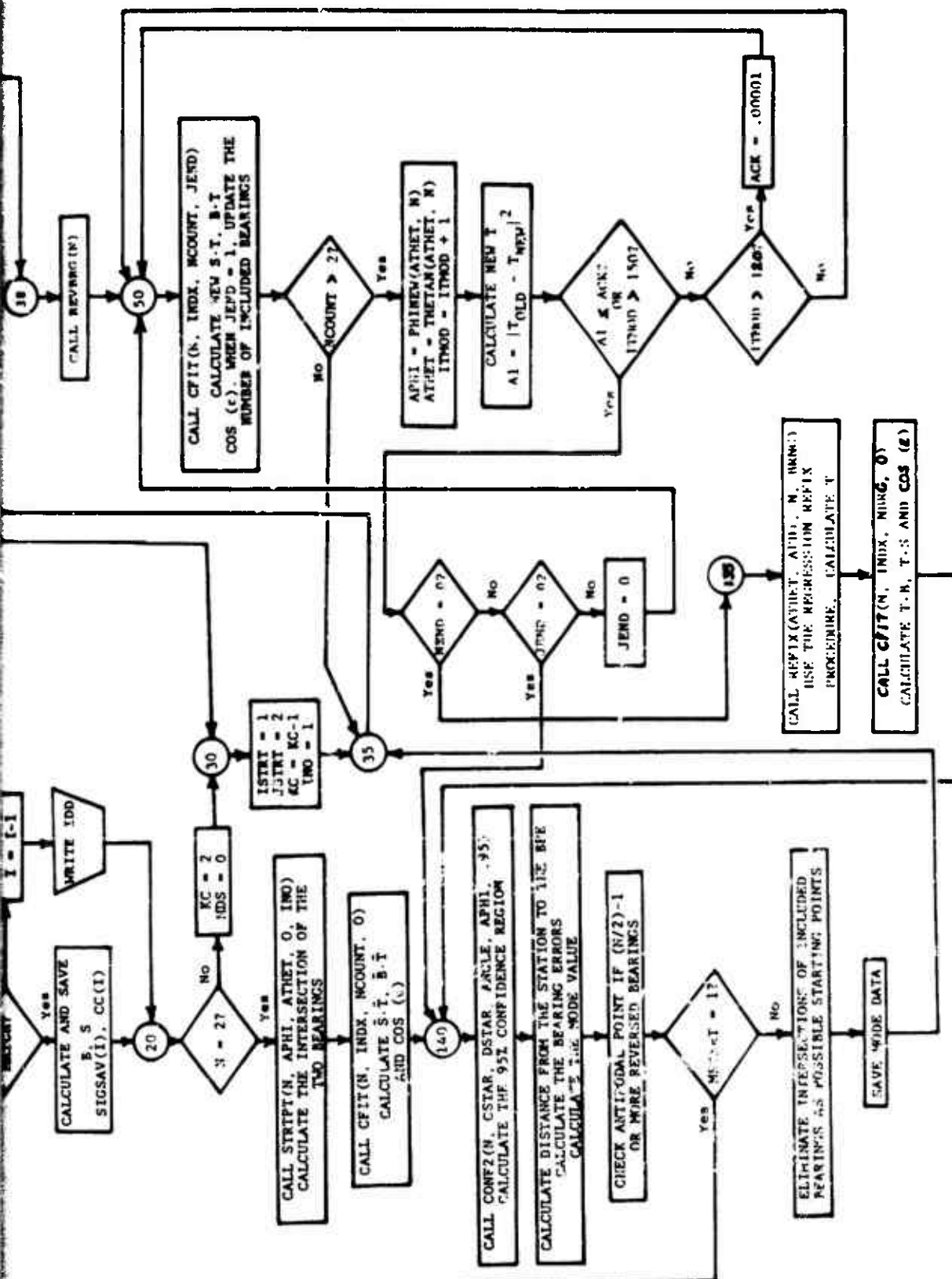
where s_0^2 is the estimate of σ_0^2 based on n data points and $F_{\alpha, 2, n-2}$ in the appropriate F distribution percentage point. In this algorithm, the σ_0^2 is assumed known so $F_{\alpha, 2, \infty}$ is used. This value is approximated by $-\log_e(\alpha)$.

The ellipse which is generated is given by the direction of the major axis and the magnitudes of the semi-major and semi-minor axes.

Appendix A

Computer Program Flow Chart





Appendix B

Computer Program Listing

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[illegible]

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[illegible]


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044100 STOP=VART-1
044200 GO 205, I=1, ISTOP
044300 IF I=1 GO 205
044400 I=I+1
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044900 I=I+1, VART
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050000 I=I+1, VART

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```

056900      FUNCTION ARCCOS(A)
056910      THIS FUNCTION CAN BE CALLED WITH VALUES OF A. IT ALLOWS THE
056920      USER TO BE SURE THAT ONE SINCE ROUND OFF ERRORS CAN
056930      GIVE SLIGHTLY NEGATIVE VALUES OF A.
056940      IF ARCCOS(A).LT.1)ARCCOS=ARCCOS(A)
056950      RETURN
056960      END
056970
056980
056990
057000
057010
057020
057030

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057400      FUNCTION ARCSIN(A)
057410      THIS FUNCTION CAN BE CALLED WITH VALUES OF A. IT ALLOWS THE
057420      USER TO BE SURE THAT ONE SINCE ROUND OFF ERRORS CAN
057430      GIVE SLIGHTLY NEGATIVE VALUES OF A.
057440      IF ARCSIN(A).LT.1)ARCSIN=ARCSIN(A)
057450      RETURN
057460      END
057470
057480
057490
057500
057510
057520
057530

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057900      SUBROUTINE ASORT(X,N,IORD)
057910      THIS SUBROUTINE SORTS THE ARRAY X AND RETURNS THE INDEX OF
057920      THE LARGEST IN IORD(1), THE SECOND LARGEST IN IORD(2), ETC.
057930      IORD IS AN INTEGER ARRAY OF DIMENSION (25)
057940      X IS A REAL ARRAY OF DIMENSION (N)
057950      N IS AN INTEGER
057960      IORD(1) IS THE INDEX OF THE LARGEST ELEMENT IN X
057970      IORD(2) IS THE INDEX OF THE SECOND LARGEST ELEMENT IN X
057980      IORD(3) IS THE INDEX OF THE THIRD LARGEST ELEMENT IN X
057990      IORD(4) IS THE INDEX OF THE FOURTH LARGEST ELEMENT IN X
058000      IORD(5) IS THE INDEX OF THE FIFTH LARGEST ELEMENT IN X
058010      IORD(6) IS THE INDEX OF THE SIXTH LARGEST ELEMENT IN X
058020      IORD(7) IS THE INDEX OF THE SEVENTH LARGEST ELEMENT IN X
058030      IORD(8) IS THE INDEX OF THE EIGHTH LARGEST ELEMENT IN X
058040      IORD(9) IS THE INDEX OF THE NINTH LARGEST ELEMENT IN X
058050      IORD(10) IS THE INDEX OF THE TENTH LARGEST ELEMENT IN X
058060      IORD(11) IS THE INDEX OF THE ELEVENTH LARGEST ELEMENT IN X
058070      IORD(12) IS THE INDEX OF THE TWELFTH LARGEST ELEMENT IN X
058080      IORD(13) IS THE INDEX OF THE THIRTEENTH LARGEST ELEMENT IN X
058090      IORD(14) IS THE INDEX OF THE FOURTEENTH LARGEST ELEMENT IN X
058100      IORD(15) IS THE INDEX OF THE FIFTEENTH LARGEST ELEMENT IN X
058110      IORD(16) IS THE INDEX OF THE SIXTEENTH LARGEST ELEMENT IN X
058120      IORD(17) IS THE INDEX OF THE SEVENTEENTH LARGEST ELEMENT IN X
058130      IORD(18) IS THE INDEX OF THE EIGHTEENTH LARGEST ELEMENT IN X
058140      IORD(19) IS THE INDEX OF THE NINETEENTH LARGEST ELEMENT IN X
058150      IORD(20) IS THE INDEX OF THE TWENTIETH LARGEST ELEMENT IN X
058160      IORD(21) IS THE INDEX OF THE TWENTYFIRST LARGEST ELEMENT IN X
058170      IORD(22) IS THE INDEX OF THE TWENTYSECOND LARGEST ELEMENT IN X
058180      IORD(23) IS THE INDEX OF THE TWENTYTHIRD LARGEST ELEMENT IN X
058190      IORD(24) IS THE INDEX OF THE TWENTYFOURTH LARGEST ELEMENT IN X
058200      IORD(25) IS THE INDEX OF THE TWENTYFIFTH LARGEST ELEMENT IN X
058210      RETURN
058220      END
058230
058240
058250
058260
058270
058280
058290

```

```

100 CONTINUE
TEMP(I)=TEMP(I)+ASLV
TEMP(I)=TEMP(I)+ASLV
100 CONTINUE

```

[illegible]

[illegible]

[illegible]

```

CSTAR=AMAX1(CSTAR,1.)
DCSTAR=AMAX1(DSTAR,1.)
RETURN
END

```

```

C++ SUBROUTINE CROSS(A,B,C)
C++ CALCULATE THE CROSS PRODUCT A CROSS B AND RETURN THE
C++ RESULT IN C
C++ FIRST INDEX REPRESENTS THE VECTOR COMPONENTS:
C++ DIMENSION 1(3),B(3),C(3)
C++ DIMENSION 2(3),A(3)*B(3)
C++ DIMENSION 3(3),A(2)*B(1)
C++ DIMENSION 4(3),A(1)*B(2)-A(2)*B(1)
C++ RETURN
C++ END

```

```

C++ CONVERT THE DMS(AA,AD,AM,AS,HH,IES,MINUTES AND SECONDS.
CONVERTS A DECIMAL ANGLE TO DEGREES,MINUTES AND SECONDS.
INTEGER AD,AM,AS
DATA NY,NS,NE,NW,'N','S','E','W' /
A=AA*57.29577951
IF(I,OPT.EQ.1)GO TO 10
IF(H=NY) I=0)IH=VS
IF(I=I)GO TO 20
IF(H=NE) I=0)IH=NS
IF(I=I)GO TO 20
IF(H=NS) I=0)IH=NW
IF(I=I)GO TO 20
J=ABS(A)-I)
AJ=60*(A-ABS(A)-I)
K=60*(A-J)
KDEL}
AM=K
ASH

```



```

076400 SUBROUTINE FLAGST (N, INC) TO 1 (INDICATING THE INTERSECTION OF
076500 SET THE VALUES TO 1 (STARTING POINT) FOR ALL INTERSECTIONS OF
076600 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
076700 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
076800 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
076900 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077000 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077100 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077200 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077300 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077400 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077500 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077600 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077700 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF
077800 CANNOT BE USED AS A STARTING POINT) FOR ALL INTERSECTIONS OF

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077900 FUNCTION FROM (SIG, SIGM, SDT) THE STANDARD DEVIATIONS OF A BEARING AS
078000 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078100 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078200 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078300 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078400 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078500 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078600 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078700 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078800 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
078900 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079000 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079100 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079200 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079300 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079400 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079500 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079600 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079700 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079800 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
079900 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
080000 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS
080100 THE FUNCTION OF THE CHANGE FROM THE STATION TO THE TARGET BEARING AS

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088200
088300
088400
088500
088600
088700
088800

```

T(2)=T(2)/DET
T(3)=T(3)/DET
PHI=ASIN(T(3))
THETA=CC(T(1))/COS(PHI)
IEI(T(2)-LT(0))THETA=-THETA
END

```

088900
089000
089100
089200
089300
089400
089500
089600
089700
089800
089900
090000
090100
090200

```

SUBROUTINE REVER3(N)
  REVERSED=1
  IREV=0
  IF (DET(20).GT(0)) THEN
    IREV=1
    REVERSED=0
  ELSE
    IREV=0
    REVERSED=1
  END IF
  DO 200 I=1, N
    DET(I)=DET(I)*I
    IF (DET(I).GT(0)) THEN
      IREV=1
    ELSE
      IREV=0
    END IF
  END DO
  IF (DET(I).GT(0)) THEN
    IREV=1
  ELSE
    IREV=0
  END IF
  RETURN
END

```

090300
090400
090500
090600
090700
090800
090900
091000
091100
091200
091300

```

FUNCTION SIGC(SQ,SIGMA,SOT)
  THIS FUNCTION CALCULATES THE STANDARD DEVIATIONS OF A BEARING AS
  A FUNCTION OF THE RANGE FROM THE STATION TO THE TARGET VECTOR.
  THE PROCEDURE USES AN APPROXIMATION TO THE RUSS RANGE CURVE.
  ITR 79-5:231
  SQ IS THE SQUARE OF THE DISTANCE FROM THE STATION TO THE TARGET.
  SOT IS THE COSINE OF THE DISTANCE. SIGM IS THE INPUT STANDARD
  DEVIATION AND IS USED AS THE LONG RANGE STANDARD DEVIATION.
  THE CONSTANT 1.826352 IS THE LONG RANGE MULTIPLIER OF THE STANDARD

```

[illegible]

[illegible]

Appendix C

Sample Output

TEST CASE NR 1

NUMBER OF BEARINGS: 6 CMT LALO: 39- 0- 0N 105- 0- 0W NUMBER OF WJDES: 1

BPE: 39-29-29N 34.6 x 30.4 (53.) 104-57- 0W

AREA: 3300.0

BEARINGS	INC	DISTANCE	ERROR
212:01	1	651.7	0.33574
103:01	1	913.7	-0.5574
161:51	1	799.5	-0.5574
333:51	1	823.3	-0.5574
261:51	1	1486.7	-0.5574
261:51	1	1364.5	-0.5574

BEARING TO CMT	DIST TO ELLIPSE	DIST TO CMT	RATIO	SEARCH AREA	INCLUDED BEARINGS	FORWARD BEARINGS	LOCK SIZE
184.6	32.0	29.1	0.911	3300.	6	6	239.405

TEST CASE NR 2

NUMBER OF BEARINGS: 7 CMT LALO: 38-18- ON 85-54- ON NUMBER OF MODES: 2

BPE: 39-27-44N 104-56-24W 86- 2-13M

AREA: 4548. 2789.92.)

AXES: 47.0 X 30.8 (55.) 41.0 X 21.6 (92.)

BEARINGS	INC	DISTANCE	ERROR	INC	DISTANCE	ERROR
212:07:0	1	653:1	0:43:0	0	795:3	75:1
107:07:5	1	914:3	-0:33:3	0	1730:1	15:0
341:51:0	1	794:3	0:0:0	0	1963:5	-0:0:4
341:51:0	0	891:5	37:34:3	1	1948:3	-0:0:4
341:51:0	0	1485:1	31:10:0	1	818:3	-0:0:4
341:51:0	1	1364:1	-15:0	1	480:3	-0:0:4
341:51:0	0	1505:5		1	605:3	-0:0:4

BEARING	ELIPSE	DIST TO CMT	RATIO DC/SE	SEARCH AREA	INCLUDES BEARINGS	FORWARDS BEARINGS	C CKGJ MAT SIZE
88.4	39.7	891.7	22.476	4548.	4	4	28.985
135.6	27.5	12.7	0.452	2789.	4	4	32.191

TEST CASE NR 3

NUMBER OF BEARINGS: 5 CMT LALO: 35-50- ON 75-48- OM NUMBER OF MODES: 2

BPE: 36-46-17N 75-56-59E 36-22- 5N 77-20-17W

AREA: 50.9 X 189. 28.8 X 238. 69.)

BEARINGS	INC	DISTANCE	ERROR	INC	DISTANCE	ERROR
PI 211.8	1	715.3	-0.6	0	768.0	-4.3
LN 224.8	0	314.2	7.3	1	374.3	0.9
MI 271.3	1	112.4	1.0	1	58.2	-179.1
PB 120.8	0	696.1	-4.6	1	653.2	0.2
	1	1186.6	0.9	1	1157.4	-2.3

BEARING TO CMT	DIST TO ELLIPSE	DIST TO CMT	RATIO DE/SE	SEARCH AREA	INCLUDED BEARINGS	FURWARD BEARINGS	CLOCKWISE SIZE
74.8	12.6	7.6	0.598	159.	3	3	154.005
68.8	28.8	79.1	2.752	238.	4	3	253.104

TEST CASE NM 4

NUMBER OF BEARINGS: 8 CMT LALO: 39-20- ON 85-50- ON NUMBER OF MILES: 5

BPE: 38-29-23N 86- 2-13W 39-26-38N 104-38-24W 41-39-51N 03- 3- 5W
 AREA: 41.0 X 27.9. 94.9 X 37.8 (90.) 75.7 X 43.7 (125.)
 AXES:

BEARINGS	INC	DISTANCE	ERRORS	INC	DISTANCE	ERRORS	INC	DISTANCE	ERRORS
107.0	0	1130.0	15.0	1	1027.7	-0.0	0	1779.1	23.4
104.0	0	1103.0	-0.0	1	982.7	-0.0	1	1903.2	-23.4
321.0	1	1098.0	-0.0	1	984.4	0.0	1	1903.2	-23.4
320.0	1	1098.0	-0.0	1	1472.0	21.2	0	1903.2	-23.4
511.0	1	1098.0	-0.0	1	1472.0	21.2	0	1903.2	-23.4
120.0	0	1098.0	-11.0	1	1472.0	21.2	0	1903.2	-23.4

BEARING TO CAT	DIST TO ELLIPSE	DIST TO CMT	RATIO DC/DE	SEARCH AREA	INCLUDED BEARINGS	FORWARD BEARINGS	COCKED MAT SIZE
125.7	30.0	13.0	0.426	2789.	4	4	32.191
88.3	94.0	880.0	9.289	11285.	3	3	9.701
213.0	43.0	237.0	5.470	10085.	3	3	1983.530

HDF/IN (06/18/79)

100	TEST	000	CASE NR 1
200	PRRRI	007	-105.00
300	BYV	125	07
400	TEST	105	07
500	PRRRI	105	07
600	BYV	105	07
700	TEST	105	07
800	PRRRI	105	07
900	BYV	105	07
1000	TEST	105	07
1100	PRRRI	105	07
1200	BYV	105	07
1300	TEST	105	07
1400	PRRRI	105	07
1500	BYV	105	07
1600	TEST	105	07
1700	PRRRI	105	07
1800	BYV	105	07
1900	TEST	105	07
2000	PRRRI	105	07
2100	BYV	105	07
2200	TEST	105	07
2300	PRRRI	105	07
2400	BYV	105	07
2500	TEST	105	07
2600	PRRRI	105	07
2700	BYV	105	07
2800	TEST	105	07
2900	PRRRI	105	07
3000	BYV	105	07
3100	TEST	105	07
3200	PRRRI	105	07
3300	BYV	105	07
3400	TEST	105	07
3500	PRRRI	105	07
3600	BYV	105	07
3700	TEST	105	07
3800	PRRRI	105	07
3900	BYV	105	07
4000	TEST	105	07
4100	PRRRI	105	07
4200	BYV	105	07
4300	TEST	105	07
4400	PRRRI	105	07
4500	BYV	105	07
4600	TEST	105	07
4700	PRRRI	105	07
4800	BYV	105	07
4900	TEST	105	07
5000	PRRRI	105	07
5100	BYV	105	07
5200	TEST	105	07
5300	PRRRI	105	07
5400	BYV	105	07
5500	TEST	105	07
5600	PRRRI	105	07
5700	BYV	105	07
5800	TEST	105	07
5900	PRRRI	105	07
6000	BYV	105	07
6100	TEST	105	07
6200	PRRRI	105	07
6300	BYV	105	07
6400	TEST	105	07
6500	PRRRI	105	07
6600	BYV	105	07
6700	TEST	105	07
6800	PRRRI	105	07
6900	BYV	105	07
7000	TEST	105	07
7100	PRRRI	105	07
7200	BYV	105	07
7300	TEST	105	07
7400	PRRRI	105	07
7500	BYV	105	07
7600	TEST	105	07
7700	PRRRI	105	07
7800	BYV	105	07
7900	TEST	105	07
8000	PRRRI	105	07
8100	BYV	105	07
8200	TEST	105	07
8300	PRRRI	105	07
8400	BYV	105	07
8500	TEST	105	07
8600	PRRRI	105	07
8700	BYV	105	07
8800	TEST	105	07
8900	PRRRI	105	07
9000	BYV	105	07
9100	TEST	105	07
9200	PRRRI	105	07
9300	BYV	105	07
9400	TEST	105	07
9500	PRRRI	105	07
9600	BYV	105	07
9700	TEST	105	07
9800	PRRRI	105	07
9900	BYV	105	07
10000	TEST	105	07

HDF/STATION (06/18/79)

100	●		
200	AB	48.9	2.5
300	VC	49.2	2.5
400	WR	45.4	2.5
500	SR	34.0	2.5
600	BR	26.0	2.5
700	WI	25.8	2.5
800	NF	36.7	2.5
900	LI	41.0	2.5
1000	PI	47.2	2.5

- 97.4	
-123.0	
-124.0	
-120.1	
-097.5	
-090.3	
-076.2	
-072.0	
-049.2	